

AD-A252 370



(2)

TECHNICAL REPORT BRL-TR-3368

BRLA REEXAMINATION OF THE
PLASTIC FLOW CRITERION FOR COPPER

NORRIS J. HUFFINGTON, JR.

DTIC
ELECTE
S JUL 07 1992 **D**
A

JULY 1992

APPROVED FOR PUBLIC RELEASE; DISTRIBUTION IS UNLIMITED.

U.S. ARMY LABORATORY COMMAND

BALLISTIC RESEARCH LABORATORY
ABERDEEN PROVING GROUND, MARYLAND

92-17565



NOTICES

Destroy this report when it is no longer needed. DO NOT return it to the originator.

Additional copies of this report may be obtained from the National Technical Information Service, U.S. Department of Commerce, 5285 Port Royal Road, Springfield, VA 22161.

The findings of this report are not to be construed as an official Department of the Army position, unless so designated by other authorized documents.

The use of trade names or manufacturers' names in this report does not constitute indorsement of any commercial product.

REPORT DOCUMENTATION PAGE			Form Approved OMB No. 0704-0188	
Public reporting burden for this collection of information is estimated to average 1 hour per response, including the time for reviewing instructions, searching existing data sources, gathering and maintaining the data needed, and completing and reviewing the collection of information. Send comments regarding this burden estimate or any other aspect of this collection of information, including suggestions for reducing this burden, to Washington Headquarters Services, Directorate for Information Operations and Reports, 1215 Jefferson Davis Highway, Suite 1204, Arlington, VA 22202-4302, and to the Office of Management and Budget, Paperwork Reduction Project (0704-0188), Washington, DC 20503.				
1. AGENCY USE ONLY (Leave blank)	2. REPORT DATE July 1992	3. REPORT TYPE AND DATES COVERED Final, Jan - Dec 91		
4. TITLE AND SUBTITLE A Reexamination of the Plastic Flow Criterion for Copper		5. FUNDING NUMBERS PR: 1L161102AH43		
6. AUTHOR(S) Norris J. Huffington, Jr.				
7. PERFORMING ORGANIZATION NAME(S) AND ADDRESS(ES) U.S. Army Ballistic Research Laboratory ATTN: SLCBR-TB-W Aberdeen Proving Ground, MD 21005-5066		8. PERFORMING ORGANIZATION REPORT NUMBER		
9. SPONSORING/MONITORING AGENCY NAME(S) AND ADDRESS(ES) U.S. Army Ballistic Research Laboratory ATTN: SLCBR-DD-T Aberdeen Proving Ground, MD 21005-5066		10. SPONSORING/MONITORING AGENCY REPORT NUMBER BRL-TR-3368		
11. SUPPLEMENTARY NOTES This report is similar to a paper with the same title presented at the Army Symposium on Solid Mechanics, 7 Nov 91.				
12a. DISTRIBUTION / AVAILABILITY STATEMENT Approved for public release; distribution is unlimited		12b. DISTRIBUTION CODE		
13. ABSTRACT (Maximum 200 words) It was found impossible to reconcile discrepancies between uniaxial and torsion test data for copper within the framework of the von Mises yield function; use of a more general function employing both the second and third invariants of the stress deviator was studied and certain limitations on use of such functions are discussed.				
14. SUBJECT TERMS constitutive models; hydrocodes; finite elements; finite strains; plastic flow; copper tensor invariants; yield function; isotropic hardening; finite element analysis			15. NUMBER OF PAGES 19	
			16. PRICE CODE	
17. SECURITY CLASSIFICATION OF REPORT UNCLASSIFIED	18. SECURITY CLASSIFICATION OF THIS PAGE UNCLASSIFIED	19. SECURITY CLASSIFICATION OF ABSTRACT UNCLASSIFIED	20. LIMITATION OF ABSTRACT UL	

INTENTIONALLY LEFT BLANK.

TABLE OF CONTENTS

	<u>Page</u>
ACKNOWLEDGMENT	v
1. INTRODUCTION	1
2. QUASI-STATIC MATERIAL PROPERTIES	1
3. CONSTITUTIVE MODELING CONSIDERATIONS	2
4. USE OF EXPERIMENTAL DATA; DISCREPANCIES	3
5. IMPLEMENTATION OF A (J_2 , J_3) THEORY	5
6. GEOMETRIC REPRESENTATION	6
7. MATERIAL STABILITY, CONVEXITY REQUIREMENTS	7
8. CONCLUDING REMARKS	8
9. REFERENCES	11
DISTRIBUTION LIST	13

DTIC COPY

Accession For	
NTIS CR&I	<input checked="" type="checkbox"/>
DTIC TAB	<input type="checkbox"/>
Unannounced	<input type="checkbox"/>
Justification	
By	
Distribution/	
Availability Codes	
Dist	Avail. or Status
A-1	

INTENTIONALLY LEFT BLANK.

ACKNOWLEDGMENT

The author benefited from several stimulating discussions with
Dr. Joseph M. Santiago, Jr., of Ballistic Research Laboratory, Aberdeen Proving Ground, MD.

INTENTIONALLY LEFT BLANK.

1. INTRODUCTION

While finite element hydrocodes can predict the contours of finitely deforming metals with reasonable accuracy, their ability to determine local strain and stress states leading to catastrophic failure by such mechanisms as adiabatic shear banding and void openings leaves much to be desired. An improvement in the ability to compute such local state histories would significantly enhance the design of warheads and predictions of armor penetration and behind-armor debris formation. In recognition that better material characterization was an essential ingredient of the desired improvement, a joint U.S. Army Ballistic Research Laboratory/Materials Technology Laboratory (BRL/MTL) program entitled Advanced Constitutive Models was initiated several years ago. This report discusses one facet of this investigation.

At BRL, it was decided to employ the DYNA3D hydrocode (Hallquist 1983) in connection with this study. This is a Lagrangian finite element computer program currently in widespread use which employs an adequate finite deformation formulation and features a choice of approximately 30 constitutive models.* However, only a few of these models are suitable for the large strain, low rate applications to be discussed in sequel. These models generally ** employ the von Mises yield condition $J_2 - k^2 = 0$, where J_2 is the second invariant of the stress deviator tensor, as the plastic potential function and use the Jaumann stress rate to account for the rigid body rotation of elements.

2. QUASI-STATIC MATERIAL PROPERTIES

For an evaluation of material constitutive behavior, it is necessary to have test data for the specific lot of material to be characterized. It was decided that this investigation would commence with a study of oxygen-free, high conductivity (OFHC) copper and MTL was tasked

* This term is used to identify a mathematical function or algorithm which determines stress tensor components from the current rate-of-deformation components, some past history data, and possibly temperature through an incremental plasticity time marching process.

** An exception is the model of D. Bammann (Johnson and Bammann 1984) as incorporated by M. Chiesa of Sandia/Livermore which is a multi-parameter micromechanically-based constitutive system which uses the Green-Naghdi stress rate.

to perform the necessary experimentation. Dr. Tusit Weerasooriya has reported (Weerasooriya and Swanson 1991) data from two types of quasi-static (isothermal) tests on annealed copper:

- (1) Uniaxial compression stress-strain curves for the range $-1.30 < \epsilon_{xx} < 0$ natural strain.
- (2) Torsional shear stress-strain data from twist tests on modified Lindholm-type thin-walled tubular specimens for the range $0 < \epsilon_{zx} < 1.4$ (tensor) shear strain. These tests were performed for two conditions of axial restraint: (a) almost total axial restraint in which case the induced axial force was recorded and (b) no applied axial restraint where the axial displacement was monitored.

Also, Dr. Weerasooriya provided the author a curve for reversed loading of a torsion specimen which permitted an assessment of the Bauschinger effect for this material.

3. CONSTITUTIVE MODELING CONSIDERATIONS

The compression test data cited above reveal that the stress-strain curve for annealed copper is nonlinear over the entire range, the elastic portion being of negligible size. For a curve of this form, a bilinear representation (such as DYNA3D Material 3) is unsatisfactory. Material 10 of the DYNA code, which permits input of up to 16 stress-strain points and interpolates linearly for intermediate values, is more appropriate but only treats isotropic hardening.

The Lindholm-type torsion specimen does not result in a homogeneous state of stress in the thin-walled test section; also, the shearing strain and plastic deformation extend into the transition section. For this reason, various investigators have resorted to 3-D finite element modeling of the whole specimen (using an assumed constitutive model) to provide a basis for interpretation of test data. Lipkin et al. (1987) reported a DYNA3D calculation using an earlier version of the Bammann constitutive model in which a twist rate high enough to predict adiabatic shear banding was used. The present author performed a DYNA3D analysis for the MTL geometry using Material 3 and found this geometry was prone to premature torsional buckling. It was recommended that a thicker wall be used since an interpretive analysis would

be required in any event. Dr. C. S. White (1990) reported an ABAQUS analysis of a geometric configuration closely corresponding to that used by Weerasooriya and has concluded that "about 78% of the twist that is applied at the grips actually goes into the deformation in the gauge section." Although White's calculations were made for a different material, it was decided to multiply Weerasooriya's shear strain data by a factor of 0.78.

4. USE OF EXPERIMENTAL DATA; DISCREPANCIES

If the compression test data are used as input to DYNA3D Material 10 and a calculation is made for an element constrained to deform uniaxially with no lateral restraints, the code predictions are (naturally) in excellent agreement with the experimental values. (It is necessary to choose an equation-of-state which permits specifying the pressure to be proportional to the volumetric strain {proportionality constant = bulk modulus}, use the hourglass viscosity type 3 {Flanagan-Belytschko (1981) with exact volume integration}, and to set the Gruneisen coefficient = 0 for an isothermal calculation.) Similarly, when torsion test data are inserted in DYNA3D Material 10 and a simple shear problem is run for an element, there is no discrepancy.

However, when compression test data are used in Material 10 to predict the stresses in an element subjected to increasing simple shear the result shown in Figure 1 is obtained, where the overprediction of the shear stress σ_{zx} is as great as 30%. Not surprisingly, the converse is also true: use of the torsion test data as input for a uniaxial compression calculation results in a significant underprediction of the axial stress σ_x as shown in Figure 2.

This phenomenon is no new discovery and has been discussed in the literature by Prager (1945), Drucker (1949), Edelman and Drucker (1951), and many others. The basic problem is that use of the von Mises condition as a loading function for work hardening materials does not closely describe the plastic deformation of many materials (even though it is widely employed for this purpose in most currently used hydrocodes). According to the authors just cited, the problem can be resolved by use of a loading function depending on both the second and third invariants of the stress deviation; i.e., a (J_2 , J_3) theory. Although such theories seem to have fallen into disuse, perhaps this approach should be considered for applications involving large strains, where the discrepancies are greatest.

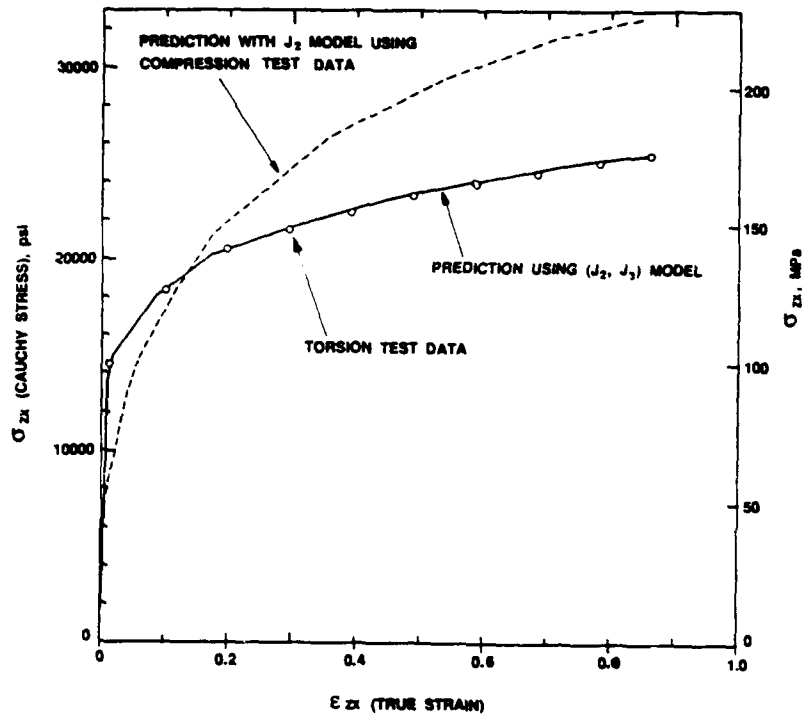


Figure 1. Simple Shear Calculation (Geometric Constraints, Material Type 10).

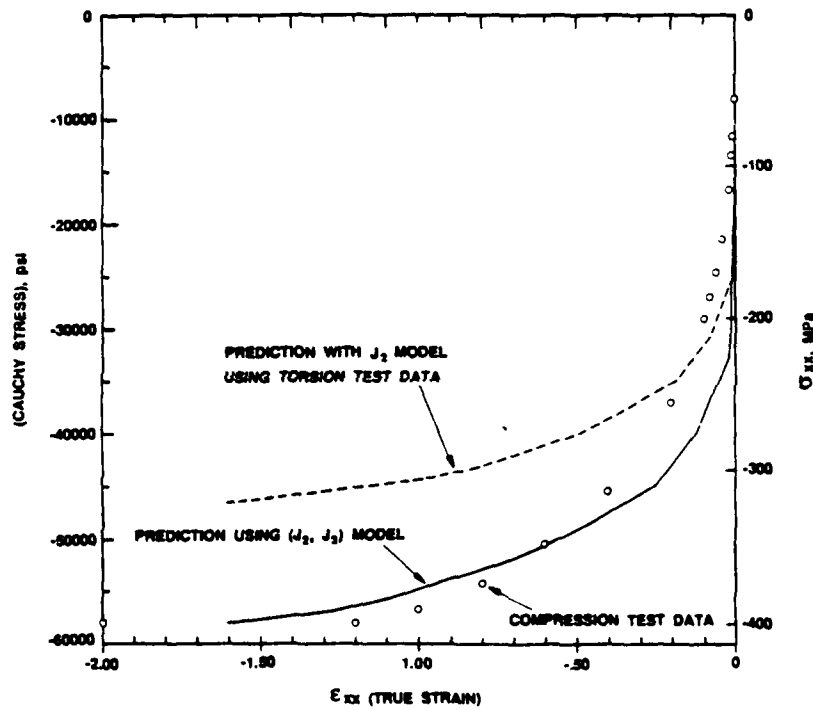


Figure 2. Uniaxial Compression Calculation (Geometric Constraints, Material Type 10).

5. IMPLEMENTATION OF A (J_2 , J_3) THEORY

It was decided to employ the function quoted by Malvern (1969):

$$f(J_2, J_3) \equiv J_2 \left[1 - \frac{c(J_3)^2}{(J_2)^3} \right] - k^2 = 0 \quad (1)$$

where

$$J_2 = \frac{1}{2} (S_{11}^2 + S_{22}^2 + S_{33}^2) + S_{12}^2 + S_{23}^2 + S_{31}^2 = \frac{1}{2} (S_1^2 + S_2^2 + S_3^2) \quad (2)$$

$$\begin{aligned} J_3 &= S_{11} S_{22} S_{33} + 2S_{12} S_{23} S_{31} - S_{11} S_{23}^2 - S_{22} S_{31}^2 - S_{33} S_{12}^2 \\ &= S_1 S_2 S_3 \end{aligned} \quad (3)$$

k = Yield stress in pure shear, variable for a work hardening material

c = nondimensional parameter to be adjusted to provide match of post yield flow data

S_{ij} = deviatoric stress components

S_k = principal deviatoric stresses

Installation of Equation 1 as the load function in DYNA3D was easily accomplished since current values of all required quantities are available in the stress evaluation subroutine. Because J_3 vanishes for pure shear, it is attractive to insert the tabular data from the shear tests in Material 10 and then determine the parameter c to provide correspondence to the compression test data for uniaxial stress calculations. For the present application, the peak stress was employed because the uniaxial stress-strain curve appears to level off at this value and the interest is in large strain plasticity. Of course, other matching criteria could be adopted for other strain ranges. Only a few trials were required to obtain $c = 2.64$; this value was used in DYNA3D to obtain the solid curves displayed in Figures 1 and 2. It may be seen in Figure 2 that there is a significant difference between experimental and predicted values in the small strain region. This is attributed to the much more rounded "knee" exhibited by copper in compression than in tension or shear.

6. GEOMETRIC REPRESENTATION

It is instructive to view the (J_2, J_3) yield function in principal stress space, where it may be recalled that the von Mises function is represented by a circular cylinder coaxial with the hydrostatic line $\sigma_1 = \sigma_2 = \sigma_3$. To achieve this, the following orthogonal coordinate transformation was made:

$$\begin{aligned}\sigma_1 &= -\frac{\tau_1}{\sqrt{6}} + \frac{\tau_2}{\sqrt{2}} + \frac{\tau_3}{\sqrt{3}} \\ \sigma_2 &= -\frac{\tau_1}{\sqrt{6}} - \frac{\tau_2}{\sqrt{2}} + \frac{\tau_3}{\sqrt{3}} \\ \sigma_3 &= \frac{2\tau_1}{\sqrt{6}} + \frac{\tau_3}{\sqrt{3}}\end{aligned}\tag{4}$$

With this transformation, the τ_3 axis coincides with the hydrostatic line and the τ_1 axis is aligned with the projection of the σ_3 axis on the deviatoric plane $\sigma_1 + \sigma_2 + \sigma_3 = 0$. Using Equations 4, one obtains

$$J_2 = \frac{1}{2} (\tau_1^2 + \tau_2^2)\tag{5}$$

$$J_3 = \frac{1}{\sqrt{6}} \left(\frac{\tau_1^3}{3} - \tau_1 \tau_2^2 \right)\tag{6}$$

and, substituting these values in Equation 1, there results

$$\begin{aligned}27\tau_2^6 + \{(81 - 36c)\tau_1^2 - 54k^2\}\tau_2^4 + \{(81 + 24c)\tau_1^4 - 108k^2\tau_1^2\}\tau_2^2 \\ + (27 - 4c)\tau_1^6 - 54k^2\tau_1^4 = 0\end{aligned}\tag{7}$$

which defines the contour of the yield function in the τ_1, τ_2 -plane.

As noted by Hill (1950), it is only necessary to compute coordinates for a 30° segment of this plane, the remainder of the locus being determined by the symmetry constraints for an

isotropic material. This locus for $c = 2.64$ is displayed in Figure 3 as a solid line. It is seen that the (J_2, J_3) surface is a fluted cylinder with the von Mises cylinder inscribed. For Material 10 which provides only for isotropic hardening, these surfaces would expand uniformly as plastic deformation progresses. It should be mentioned that DYNA3D presently uses only the Krieg-Key radial return algorithm (Krieg and Key, 1976) to return the stress state to the updated yield surface each cycle. This algorithm is clearly more appropriate for the J_2 surface than for the (J_2, J_3) function. However, it is still attractive due to its simplicity and the assurance that the yield surface will be intersected, although for the latter it corresponds to a nonassociated flow rule.* A "normal return" algorithm would seem preferable or perhaps the recently published Nemat-Nasser algorithm (Nemat-Nasser 1991), but either of these would require more extensive calculations per cycle. These matters may be academic in view of the following discussion.

7. MATERIAL STABILITY, CONVEXITY REQUIREMENTS

It is clear from inspection that the $c = 2.64$ loading function in Figure 3 violates the requirement that the surface be convex, as deduced by Drucker (1951, 1959) from his postulates for material stability. Although the author has made a considerable number of computer runs using this value of c with no evidence of instability, these cases were not designed to test all types of loading and unloading. Thus, it is appropriate to ask: Can any (J_2, J_3) load function satisfy the convexity requirement? The answer is yes and the limitation on c is determined by evaluating the curvature of the load function at its point of tangency to the von Mises circle. Thus, at $\tau_1 = 0$,

$$\frac{d^2\tau_2}{d\tau_1^2} = \frac{1 - \frac{4}{3}c}{\sqrt{2} k} \quad (8)$$

This happens to be the exact curvature since $\frac{d\tau_2}{d\tau_1} = 0$ at this location.

* This consideration does not invalidate the results displayed in this report since, for both pure shear and uniaxial loading, radial and normal return coincide.

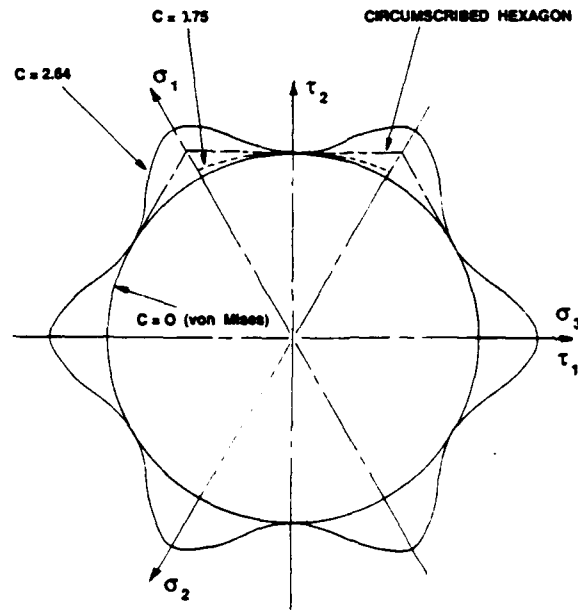


Figure 3. Axial View of Yield Loci.

Therefore, the transition in sign of the curvature occurs at $c = 0.75$ and the load function is everywhere convex for this or lesser values of c . The form of the load function for $c = 0.75$ is plotted in the upper sextant of Figure 3. Unfortunately, for this value of c , the percentage difference between the maximum and minimum radii of the load function is only about 6%. It can now be seen that the greatest percentage difference in radii consistent with convexity requires a load function in the form of a hexagon circumscribed about the von Mises circle, as shown in the upper portion of Figure 3. Even for this form of yield function (with its attendant analytical complexities), the percentage difference in radii is only 15.5%, about half of what is needed to reconcile the discrepancies associated with experimental data.

8. CONCLUDING REMARKS

It is realized that other investigators (Shrivastava, Jonas, and Canova 1982; Asaro and Needleman 1985; Weerasooriya and Swanson 1991) have treated the subject matter of this paper by micromechanical modeling and have attributed the cited discrepancies to formation

of texture. However, the end objective of such research does not appear to be the identification of load functions appropriate to classical plasticity.

This author is convinced that material behavior must be modeled within the framework of continuum mechanics if efficient, large-scale computations are to yield valid results for engineering purposes. While the study reported herein may seem inconclusive, it is believed that use of a nonconvex loading function may be permissible for certain classes of problems, especially if the computer program is modified to test on the sign of the plastic work increment in each element and to terminate the calculations if a negative work increment is predicted. It is very desirable to resolve this matter before proceeding to studies of strain-rate effects, stress rate models, etc., where uncertainties regarding the loading function may obscure interpretation of other types of experimental results.

It should be mentioned that the author has added a new subroutine to a research version of DYNA3D which includes many effects previously not present in a single material model, specifically:

- (1) Finite elastoplastic straining.
- (2) Mixed kinematic/isotropic hardening, as recommended by Hodge (1957).
- (3) Arbitrary shape of uniaxial stress-strain function through input of tabular data.
- (4) Choice of stress rate formulation (Jaumann [1905] or Green and Naghdi [Green and Naghdi 1965; Green and McInnis 1967]).
- (5) Choice of several equation-of-state models.

It is planned to use this tool to study finite plasticity for various proportional and nonproportional loading paths.

INTENTIONALLY LEFT BLANK.

9. REFERENCES

- Asaro, R. J., and A. Needleman. "Texture Development and Strain Hardening in Rate Dependent Polycrystals." Acta Metall., vol. 33, p. 923, 1985.
- Drucker, D. C. "Relation of Experiments to Mathematical Theories of Plasticity." Journal of Applied Mechanics, vol. 16, no. 4, pp. 349–357, 1949.
- Drucker, D. C. "A More Fundamental Approach to Plastic Stress-Strain Relations." Proc. First U.S. Natl. Cong. Appl. Mechanics, ASME, pp. 487–491, 1951.
- Drucker, D. C. "A Definition of Stable Inelastic Material." Journal of Applied Mechanics, vol. 26, pp. 101–106, 1959.
- Edelman, F., and D. C. Drucker. "Some Extensions of Elementary Plasticity Theory." Journal of the Franklin Institute, vol. 251, pp. 581–605, 1951.
- Flanagan, D. P., and T. Belytschko. "A Uniform Strain Hexahedron and Quadrilateral and Orthogonal Hourglass Control." Int. J. Numer. Meths. Eng., vol. 17, pp. 679–706, 1981.
- Green, A. E. and P. M. Naghdi. "A General Theory of an Elastic-Plastic Continuum." Arch. Rat. Mech. Anal., vol. 18, pp. 251–281, 1965.
- Green, A. E. and B. C. McInnis. "Generalized Hypo-Elasticity." Proc. Roy. Soc. Edinburgh, A57, p. 220, 1967.
- Hallquist, J. O. "Theoretical Manual for DYNA3D." UCID-19401, University of California, Lawrence Livermore National Laboratory, 1983.
- Hill, R., "The Mathematical Theory of Plasticity." Oxford, p. 18, 1950.
- Hodge, P. G., Jr. Discussion of Prager (1956), Journal of Applied Mechanics, vol. 24, no. 3, pp. 482–483, 1957.
- Jaumann, G. "Grundlagen der Bewegungslehre." Leipzig, 1905.
- Johnson, G. C., and D. J. Bammann. "A Discussion of Stress Rates in Finite Deformation Problems." Int. J. Solids Structures, vol. 20, no. 8, pp. 725–737, 1984.
- Krieg, R. D. and S. W. Key. "Implementation of a Time Independent Plasticity Theory Into Structural Computer Programs." Constitutive Equations in Viscoplasticity, ASME, AMD vol. 20, 1976.
- Lipkin, J., M. L. Chiesa, and D. J. Bammann. "Thermal Softening of 304L Stainless Steel: Experimental Results and Numerical Simulations." Proc. of IMPACT'87, Bremen, Germany, 1987.

- Malvern, L. E. "Introduction to the Mechanics of a Continuous Medium." Prentice-Hall, p. 355, 1969.
- Nemat-Nasser, S. "Rate-Independent Finite-Deformation Elastoplasticity: A New Explicit Constitutive Algorithm. Mechanics of Materials, vol. 11, No. 3, pp. 235-249, 1991.
- Prager, W. "Strain Hardening Under Combined Stresses." J. Appl. Physics, vol. 16, pp. 837-840, 1945.
- Shrivastava, S. C., J. J. Jonas, and G. Canova. "Equivalent Strain in Large Deformation Torsion Testing: Theoretical and Practical Considerations." J. Mech. Phys. Solids, vol. 30, pp. 75-90, 1982.
- Weerasooriya, T., and R. A. Swanson. "Experimental Evaluation of the Taylor-Type Polycrystal Model for the Finite Deformation of an FCC Metal (OFHC Copper)." MTL TR 91-20, U.S. Army Materials Technology Laboratory, 1991.
- White, C. S. "An Analysis of the Torsion Specimen Used in Constitutive Modeling." Presented at the Eighth Army Conference on Applied Mathematics and Computing, Cornell University, 1990.

<u>No. of Copies</u>	<u>Organization</u>	<u>No. of Copies</u>	<u>Organization</u>
2	Administrator Defense Technical Info Center ATTN: DTIC-DDA Cameron Station Alexandria, VA 22304-6145	1	Commander U.S. Army Tank-Automotive Command ATTN: ASQNC-TAC-DIT (Technical Information Center) Warren, MI 48397-5000
1	Commander U.S. Army Materiel Command ATTN: AMCAM 5001 Eisenhower Ave. Alexandria, VA 22333-0001	1	Director U.S. Army TRADOC Analysis Command ATTN: ATRC-WSR White Sands Missile Range, NM 88002-5502
1	Commander U.S. Army Laboratory Command ATTN: AMSLC-DL 2800 Powder Mill Rd. Adelphi, MD 20783-1145	1	Commandant U.S. Army Field Artillery School ATTN: ATSF-CSI Ft. Sill, OK 73503-5000
2	Commander U.S. Army Armament Research, Development, and Engineering Center ATTN: SMCAR-IMI-I Picatinny Arsenal, NJ 07806-5000	(Class. only) 1	Commandant U.S. Army Infantry School ATTN: ATSH-CD (Security Mgr.) Fort Benning, GA 31905-5660
2	Commander U.S. Army Armament Research, Development, and Engineering Center ATTN: SMCAR-TDC Picatinny Arsenal, NJ 07806-5000	(Unclass. only) 1	Commandant U.S. Army Infantry School ATTN: ATSH-CD-CSO-OR Fort Benning, GA 31905-5660
1	Director Benet Weapons Laboratory U.S. Army Armament Research, Development, and Engineering Center ATTN: SMCAR-CCB-TL Watervliet, NY 12189-4050	1	WL/MNOI Eglin AFB, FL 32542-5000
(Unclass. only) 1	Commander U.S. Army Rock Island Arsenal ATTN: SMCRI-TL/Technical Library Rock Island, IL 61299-5000		<u>Aberdeen Proving Ground</u>
1	Director U.S. Army Aviation Research and Technology Activity ATTN: SAVRT-R (Library) M/S 219-3 Ames Research Center Moffett Field, CA 94035-1000	2	Dir, USAMSAA ATTN: AMXSY-D AMXSY-MP, H. Cohen
1	Commander U.S. Army Missile Command ATTN: AMSMI-RD-CS-R (DOC) Redstone Arsenal, AL 35898-5010	1	Cdr, USATECOM ATTN: AMSTE-TC
		3	Cdr, CRDEC, AMCCOM ATTN: SMCCR-RSP-A SMCCR-MU SMCCR-MSI
		1	Dir, VLAMO ATTN: AMSLC-VL-D
		10	Dir, USABRL ATTN: SLCBR-DD-T

No. of	
<u>Copies</u>	<u>Organization</u>
2	Director DARPA ATTN: J. Richardson LTC J. Beno 1400 Wilson Blvd. Arlington, VA 22209-2308
1	Defense Nuclear Agency ATTN: MAJ James Lyon 6801 Telegraph Road Alexandria, VA 22192
4	U.S. Army Research Office ATTN: I. Ahmad K. Iyer J. Wu Technical Library P.O. Box 12211 4300 Miami Blvd. Research Triangle Park, NC 27709
3	U.S. Army MICOM ATTN: AMSMI-RD-ST-WF, Lynn Craft Donald Lovelace Michael Schexnayder Redstone Arsenal, AL 35898-5250
2	U.S. Army Belvoir RD&E Center ATTN: S. G. Bishop C. Kominos Fort Belvoir, VA 22060-5166
5	Commander U.S. Army Armament Research, Development, and Engineering Center ATTN: T. Davidson V. Lindner J. Pearson E. Baker Technical Library Picatinny Arsenal, NJ 07806-5000

No. of	
<u>Copies</u>	<u>Organization</u>
6	U.S. Army Materials Technology Laboratory ATTN: SLCMT-MRT, S.C. Chou C.S. White J. McLaughlin T. Weerasooriya D. Dandekar Technical Library Watertown, MA 02172-0001
2	Naval Weapons Center ATTN: Don Thompson, Code 3268 Technical Library China Lake, CA 93555
2	Commander Naval Surface Warfare Center ATTN: William Mock Technical Library Dahlgren, VA 22448-5000
7	Commander Naval Surface Warfare Center ATTN: P. C. Huang, G-402 J. P. Matra P. Walter L. Mensi F. J. Zerilli H. Chen, U-12 Technical Library 10901 New Hampshire Ave. Silver Spring, MD 20903-5000
1	Sandia National Laboratories ATTN: D. Bammann J. Lipkin Livermore, CA 94550
1	Air Force Wright Laboratories Materials Laboratory ATTN: Dr. T. Nicholas Wright-Patterson AFB, OH 45433
3	Air Force Armament Laboratory ATTN: AFATL/DLJW, W. Cook AFATL/MNW, J. Foster Technical Library Eglin AFB, FL 32542

No. of	
<u>Copies</u>	<u>Organization</u>
4	Director Sandia National Laboratories ATTN: J.M. McGlaun P. Yarrington E. Hertel Technical Library P.O. Box 5800 Albuquerque, NM 87185
5	Director Los Alamos National Laboratory ATTN: G.E. Cort, F663 T.F. Adams, F663 D. Mandell, F663 R. Karpp, MS J960 J. Chapyak Technical Library P.O. Box 1663 Los Alamos, NM 87545
4	Director Lawrence Livermore National Laboratory ATTN: R. Tipton, L-35 R. Whirley, L-122 R. Christensen, L-35 Technical Library P.O. Box 808 Livermore, CA 94550
1	Aerojet Precision Weapons Dept. 5131/T-W ATTN: J. Carleone 1100 Hollyvale Azusa, CA 91702
3	Dyna East Corporation ATTN: P.C. Chou R. Ciccarelli W. Flis 3201 Arch Street Philadelphia, PA 19104
3	Southwest Research Institute ATTN: C. Anderson A. Wenzel U. Lindholm P.O. Drawer 28255 San Antonio, TX 78228-0255

No. of	
<u>Copies</u>	<u>Organization</u>
1	Alliant Techsystems, Inc. ATTN: G.R. Johnson MN 48-2700 7225 Northland Drive Brooklyn Park, MN 55428
1	S-Cubed ATTN: R. Sedgwick P.O. Box 1620 La Jolla, CA 92038-1620
2	Orlando Technology, Inc. ATTN: D. Matuska J. Osborn P.O. Box 855 Shalimar, FL 32579
1	Livermore Software Technology Corp. ATTN: John O. Hallquist 2876 Waverly Way Livermore, CA 94550
1	Rensselaer Polytechnic Institute ATTN: Professor E. Krempf Troy, NY 12181
3	Brown University Division of Engineering ATTN: Professor R. Clifton Professor H. Kolsky Professor R. Asaro Providence, RI 02912
1	Carnegie-Mellon University Department of Mathematics ATTN: Dr. M.E. Gurtin Pittsburgh, PA 15213
2	The Johns Hopkins University ATTN: Professor W. Sharpe Professor J. F. Bell 34th and Charles Streets Baltimore, MD 21210
1	University of California at San Diego Department of Mechanical and Aerospace Engineering ATTN: Professor S. Nemat-Nasser La Jolla, CA 92093

No. of

Copies Organization

- 1 University of Delaware
Department of Mechanical and Aerospace
Engineering
ATTN: Professor J. Vinson
Newark, DE 19711
- 2 University of Florida
Department of Engineering Science and
Mechanics
ATTN: Professor L. Malvern
Professor D. Drucker
Gainesville, FL 32601
- 1 Virginia Polytechnic Institute and State
University
Department of Engineering Science and
Mechanics
ATTN: Professor C. W. Smith, Jr.
Blacksburg, VA 24061

USER EVALUATION SHEET/CHANGE OF ADDRESS

This laboratory undertakes a continuing effort to improve the quality of the reports it publishes. Your comments/answers below will aid us in our efforts.

1. Does this report satisfy a need? (Comment on purpose, related project, or other area of interest for which the report will be used.)

2. How, specifically, is the report being used? (Information source, design data, procedure, source of ideas, etc.)

3. Has the information in this report led to any quantitative savings as far as man-hours or dollars saved, operating costs avoided, or efficiencies achieved, etc? If so, please elaborate.

4. **General Comments.** What do you think should be changed to improve future reports? (Indicate changes to organization, technical content, format, etc.)

BRL Report Number BRL-TR-3368 Division Symbol

Check here if desire to be removed from distribution list.

Check here for address change.

Current address: Organization _____
Address _____

DEPARTMENT OF THE ARMY
Director
U.S. Army Ballistic Research Laboratory
ATTN: SLCBR-DD-T
Aberdeen Proving Ground, MD 21005-5066

OFFICIAL BUSINESS

NO POSTAGE
NECESSARY
IF MAILED
IN THE
UNITED STATES

BUSINESS REPLY MAIL

FIRST CLASS PERMIT No 0001, APG, MD

Postage will be paid by addressee.

**Director
U.S. Army Ballistic Research Laboratory
ATTN: SLCBR-DD-T
Aberdeen Proving Ground, MD 21005-5066**

